Q.P.	Code:	20ME0305	
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R	Reg. No:]			
	SIDDE	IART	H INS	TITU	ΤΕΟ	F ENG	GINE	ERIN	G & '	ГЕСН	INOL	- OGY:: P	UTTU	R	
						(AU	TON	OMOL	JS)						
	B. Tee	ch II Y	ear l	Seme	ster	Supp	leme	ntary	Exan	ninati	ions I	Novembe	ər-202	2	
					THE				CERI	NG					
т	ima: 2 haura				(1)	lechai	nical I	zngine	ering)				Mox N	Acul	a: 60
1	inte. 5 nours								•		1 \		Max. N	viark	S: 00
				(Ans	swer a	ll Five	Unit	$\frac{55x1}{T}$	2 = 6	0 Mar	ks)				
1	a Derive a	n evnr	ession	for m	inimu	m	UNI rk rec	1-1	for tu	o star	te reci	inrocating	air	12	6M
1	a Derive a compress	sor wit	h perfe	ect inte	er-coo	ling a	nd neg	glect c	learan	ce vol	ume.	iprocating	an		UIVI
	b What are	the va	rious o	classif	ication	ns of a	ir con	npress	ors.]	L2	6M
							0]	R							
2	Air from a	n initi	al con W1.25	=	of 2	50C a	and I	bar a	abs is	comp	toan	d in 2 sta	age	L3	12M
	bar abs. Est	imate 1	the mi	nimun	n work	requi	red a	nd hea	t rejec	ted in	the in	tercooler	per		
	kg of air. As	ssume	CP=1.	05KJ/	Kg an	d R=0	.29KJ	Kg K							
	Table Mat Back P						UNI	Г-II							
3	Brief the wo	orking	of Bra	yton (Cycle v	with th	e helj	p of p-	v diag	ram a	nd T-s	s diagram.		L2	12M
4	In an oil ga	s turbi	ne inst	allatio	n air	is take	en at	K Ibar ai	nd 300	C Th	e air is	s compres	sed	L3	12M
	to 4 bar and	then h	neated	by bu	rning 1	the oil	to a t	temper	ature	of 500	00C.If	the air flo	OWS		LAIVE
	at the rate of	f 90K	g/Minu	ite, fir	nd the	powe	r deve	eloped	by th	e plan	ıt. Tak	te γ for air	r as		
	1.4 and Cp a	as 1KJ	/KgK.			1		TTT							
5	a Evolaina	bout s	uner c	aturate	ed flov	y in no		with 1	neat cl	etch	Andr	enresent i	n	12	8 M
2	H-S diag	ram.	super s	aturat	20 110 0	v III IIC	JZZICS	with 1	icat sr	cten.	And D	epresent n	u .		OIVI
	b What are	the ef	fects o	f frict	ion on	flow	throug	gh noz	zle?					L2	4M
	D		`				0]	R				2 1 3 6			1035
6	6 Dry saturated steam at a pressure of 11 bar is Expanded in a nozzle to 2 bar. If the flow is isentropic, determine, (i) the threat velocity (ii) exit velocity (iii) ratio of								the	L3	12M				
	cross section	nal are	a from	exit to	o throa	at.	at ve	locity	(1) 0.	xit ve	locity	(III) Tatio	01		
							UNI	Γ -ΙV							
7	a Draw and	ł expla	in the	veloci	ity tria	ngle o	f imp	ulse tu	rbine.					L2	6M
	b Derive a	n expre	ession	for wo	ork doi	ne in i	mpuls	se turb	ine.					L2	6M
8	Draw the c	ombin	ed vel	ocity	triano	e of	Parso	K n's rea	action	turbi	ne and	t explain	the	L2	12M
U	salient featu	res.	ea ver	oeny	ti iung	01	urse.		ietion	turon	ile une	a explain	the		
							UNI	Г-V							
9	A test on	a_sing	gle cy	linder	4 st	roke	Otto	cycle	engir	ne yie	elds tl	he follow	ving	L3	12M
data:950Nm Torque, 7.6 bar mean effective pressure, 280mm bore, 305mm stroke,															
	Determine:	(i) Ind	icated	therm:	al Effi	ciency	on w 7 (ii) M	Mecha	nical e	efficie	nev.	42000NJ/1	ng.		
							0	R							
10	Define Brak	e Pow	er, Ind	icated	Powe	r. Exp	lain v	various	meth	ods of	f Meas	suring the	m.	L2	12M
						* *	* EN	D ***							
							LIN	\mathbf{D}							